

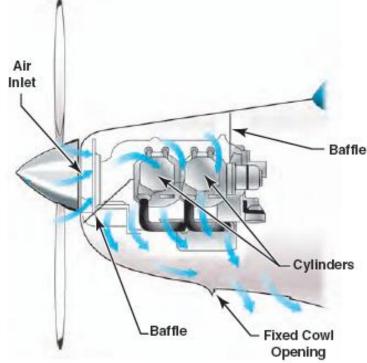




AS0403 Aircraft Systems & Instruments

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UNIT – I

AIRCRAFT SYSTEM

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Hydraulic System

- Classified into two major sections
 - Power Section
 - > Actuating Section
- Power Section :
 - Provides fluid flow
 - Regulates and limits pressure
 - > Carries fluid to various selector values in the system

Hydraulic System

Actuating Section

Contains various operating units such as wing flaps, landing gear, brakes, boost systems and steering mechanisms.

Power Section may be further classified as
> Open System
> Closed System



- It has fluid flow but no appreciable pressure in the system whenever the actuating mechanisms are idle.
- Selector valves are always connected in series with each other, whereby the pressure line goes through each selector valve.
- Pressure developed by the pump is controlled by one of the three valves:
 - > Open-center valve
 - Power-control valve
 - Pump-control valve



 It develops no pressure except when a mechanism is being operated; the pressure is then metered by selector valve and limited by a relief valve.

Advantages:

- It does not require expensive or complicated pressure regulators.
- The power pump can be a simple gear pump, although a fixed – displacement piston pump may be used



Disadvantages:

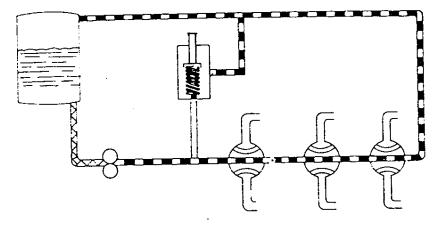
Operation of only one subsystem at a time is possible without interference from other systems.

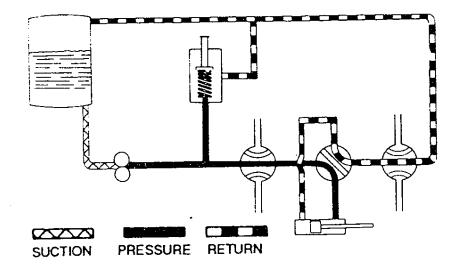
Transport-category aircraft require more complex systems, which may have several units operating at the same time.

Uses:

> Light, general aviation aircraft.

Basic Open-Center System







 It directs fluid flow to the main system manifold and builds up pressure in that portion of the system that leads to all selector valves.

- Two basic types:
 - Constant-volume pump and a pressure regulator
 - Variable-volume pump



Constant-volume pump and a pressure regulator:

≻To control pressure at working range .

≻To unload the pump when there is no flow requirement.

≻Pressure builds up in the system manifold.

Closed System

Variable-volume pump:

Directs the flow to the system manifold

- Output of the valve is controlled by an integral control valve.
- The valve reduces the pump flow to zero when no units are operating in the system and pressure is built up in the storage chambers, called accumulators.

Closed System

- Any number of subsystems may be incorporated in a closed system
- Selector valves are arranged in parallel
- Based on system pressure the fluid is directed either to the system or to the reservoir by the pressure regulator.
- System pressure is maintained between kick-out and kick-in settings of the regulator when the actuating mechanisms are not in operation.

Closed System

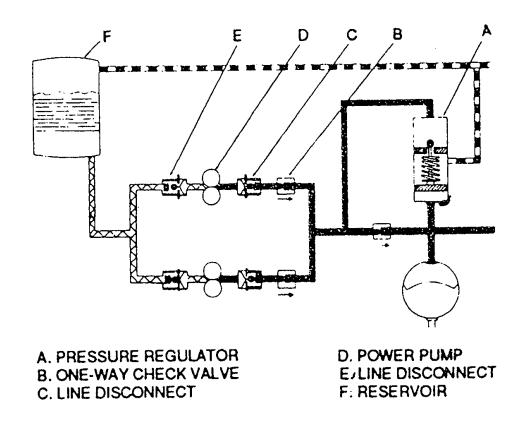
• Accumulator: It stores fluid under pressure, Stabilizes system pressure, ensures smooth operation of the regulator.

• Relief valve safeguards the system if the regulator fails.

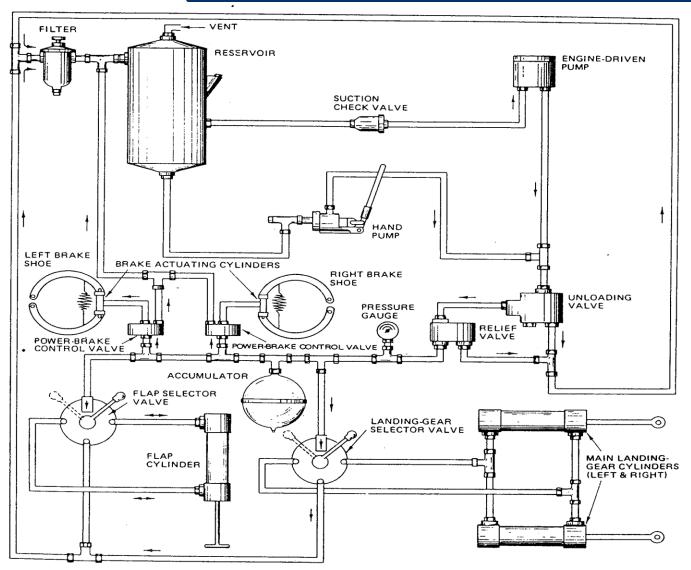
 Multiple power pumps are used in multiengine aircrafts, where they can be driven by separate engines.

Closed System with Dual Pumps

Closed System with Dual Pumps

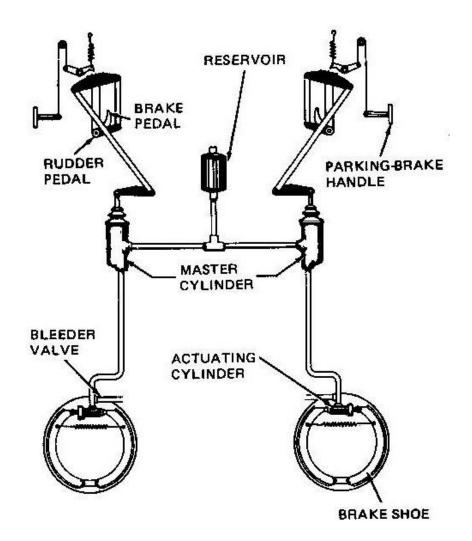


Closed- Center System

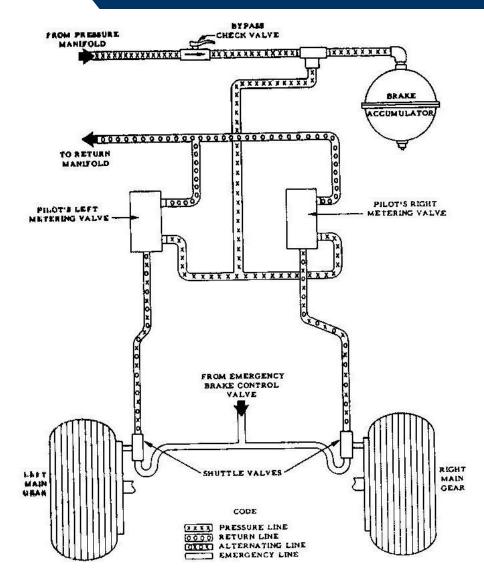


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Independent Brake System

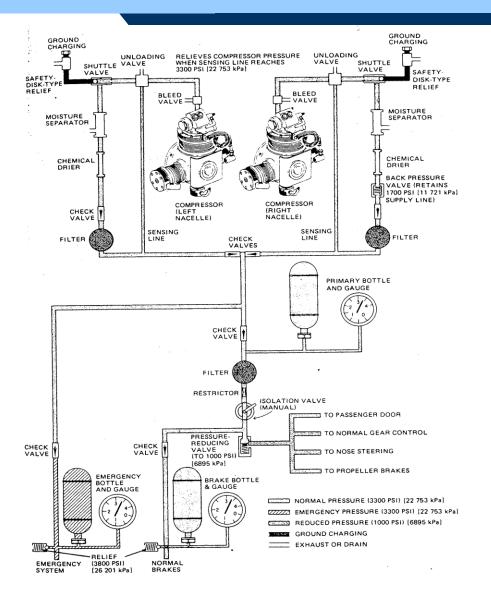


Power Boost Brake System



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Pneumatic System

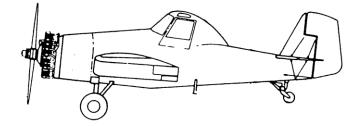


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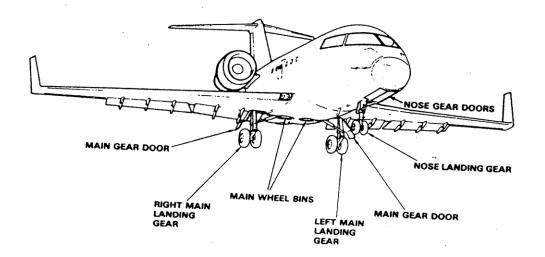
Aircraft Landing-Gear System

Configuration

Conventional Geared Aircraft



Tricycle Landing Gear



Classification

Non - Absorbing Landing Gear

- Rigid Landing Gear
- Shock-Cord Landing Gear
- Spring-Type Gear

Shock-Absorbing Landing Gear

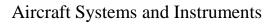
- Spring Oleo
- Air- Oleo
- Fixed Gear
- Retractable Gear
- Hulls and Floats

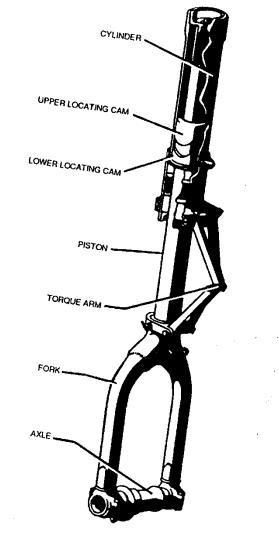
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Landing-Gear Components

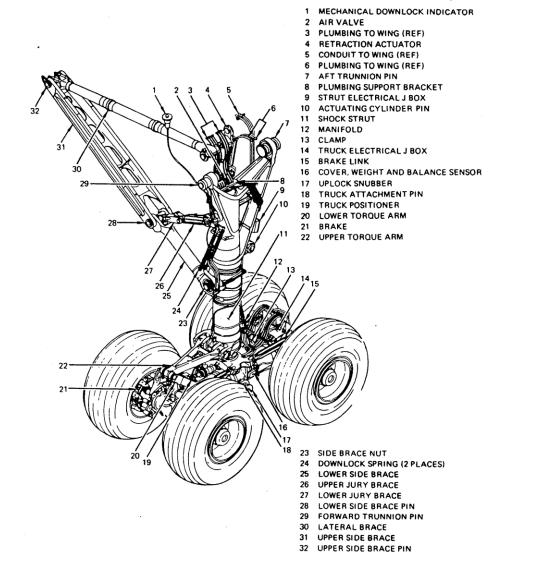
- Trunnion
- Struts
- Torque Links
- Truck or **Bogie**
- Drag Link or Drag Strut
- Side Brace Link or Side Strut
- Overcenter Link or Downlock
- Swivel Gland
- Shimmy Dampers

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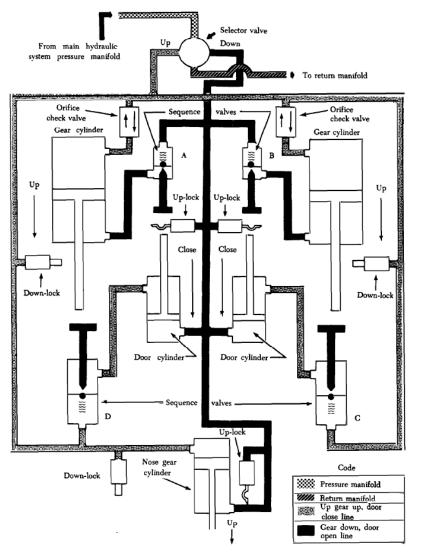




Landing-Gear Components contd..

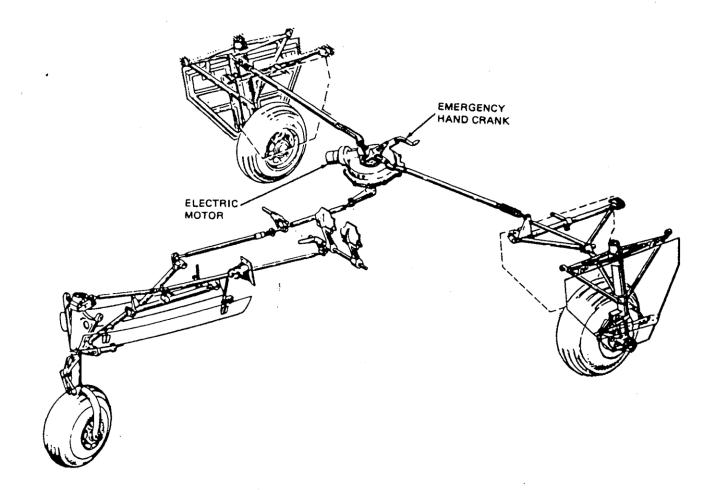


Hydraulic Landing Gear Retraction System



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Landing Gear System (Retractive Mechanism)



Electrically Operated Landing –Gear System

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